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Aerodynamic optimal design of wind turbine blades using genetic algorithm

ABSTRACT

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Wind power has been widely considered and utilized in recent years as one of the most promising renewable energy sources. In the current research study, aerodynamic analysis of the upwind three-bladed horizontal axis turbine is carried out using blade element momentum theory (BEM), and a genetic algorithm (GA) is applied as an optimization method. Output power generation is considered as an objective function, which is one of the most common choices of objective function. The optimization variables also involve chord and twist distribution variations and the placement of the airfoil sections along the blade length. The optimal blade shape is investigated for the maximum output power at a specific wind speed, rotor diameter and airfoil profile. The modified BEM results are compared with an experimental measurement indicating reliable results. The results show that considering placement of the airfoils as design variables in addition to chord and twist rate has a significant effect on the optimal output power. Finally, the objective function (output power) is improved by 10% compared to the base function.

Keywords: Aerodynamic, BEM & GA, Optimization, Output Power, Wind turbine blade.

1. Introduction

Recently, the wind power growth rate has been investigated. Wind turbine technology is one of the methods to apply this renewable resource in order to generate environmentally friendly electrical energy. These technologies are complex systems whose design requires integration the many of engineering disciplines. including aerodynamics, structures, controls and electrical engineering. The main aim of most wind turbines is to capture maximum energy from the wind energy. Therefore, it is necessary to know what the best design parameters for each component of the wind turbine would be [1]. Generally, wind turbine design techniques try

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to emphasize and address the optimal blade shape and optimal control, and increase sitespecific performance to improve wind turbine efficiency and life cycle. Aerodynamic design and structural design are two stages of a wind turbine blade design process, which can both be seen as optimization problems. Typically, the design goal is maximum aerodynamic efficiency at a specific wind speed. Aerodynamic design optimization includes the selection of an airfoil family and its placement, and the chord and twist distributions along the blade length [1]. There are different aerodynamic methods that have been developed for wind turbines, such as blade element momentum (BEM), vortex methods [2], dynamic stall models [3], CFD [4] and experimental methods. BEM method, presented by Glauert [5], enables the calculation of the steady loads and power.

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This method with some modifications often predicts the performance of wind turbines with acceptable accuracy, and also has a cheap computational run time. Many studies have been conducted on BEM method for the design and evaluation of wind turbine rotor performance [6-11].

Furthermore, genetic algorithm (GA) has successfully implemented been in aerodynamic optimization problems for horizontal axis wind turbines [12]. Liu et al. [13] investigated the linearization of the chord and twist angle radial profiles for a fixed-pitch, fixed-speed horizontal axis wind turbine using the BEM theory to calculate the power performance of the blade. They considered the highest AEP for a particular wind speed as the optimization criterion with constraints on the upper limit of power output of the wind turbine. Ceyhan [14] developed an aerodynamic design and optimization tool for horizontal axis wind turbines using both the BEM theory and GA. Blades were optimized for the maximum power output for a given wind speed. Chord, twist angle and a fixed number of sectional airfoil profiles were considered as design variables. Polat and Tuncer [15] developed an aerodynamic shape optimization methodology based on genetic algorithm and blade element momentum theory for rotor blades of horizontal axis wind turbines. They maximization of power studied the production at a specific wind speed, rotor speed and rotor diameter. The sectional chord length, twist and blade profiles at the root, middle and tip regions of the blade were considered as design variables. Liu et al. [16] studied an optimization model using the extended compact genetic algorithm (ECGA) for rotor blades of a 1.3 MW stallregulated wind turbine with an objective function to satisfy the maximum annual energy output on specific winds. Mendez and Greiner [17] illustrated the optimal wind blade shape using a genetic algorithm. This work focused on optimization of wind turbine application in particular wind conditions, which might differ from the standard classes according to the IEC classification.

Most of the research has reported on and studied the optimization of twist and chord rate. In the current research study, the placement of different types of airfoil along the blade is considered. Blade optimization of the Vestas 660 kW wind turbine is investigated from an aerodynamic point of view. The methodology is implemented within an integrated GA method and modified BEM developed by computer code for this complex wind turbine blade. The optimization is focused on maximizing power generation, determined by optimal chord and twist rates and the optimal placement of the airfoils along the blade length.

Nomenclature

a	Axial Induction Factor
a'	Tangential Induction Factor
С	Chord (m)
CD	Drag Coefficient
CL	Lift Coefficient
C _N	Normal Force Coefficient
CT	Tangential Force Coefficient
f(x)	objective function
$g_j(x)$	inequality constraint
m	number of constraints
n	number of design variables
Р	Power (KW)
r	Radial distance from root (m)
Т	Twist (deg)
Vo	Wind Speed (m/s)
Va	Axial Induced Velocity (m/s)
Vrot	Rotational Induced Velocity
	(m/s)
x	design variable
x_i^L	Lower limits
x_i^U	Upper limits

Greek Symbols

α	Angle of Attack (deg)	
α	Angle of Attack (deg)	

α_{pitch}	Pitch Angle (deg)
Φ	Inflow Angle (deg)
Σ	Solidity

Ω Rotational Velocity (rpm)

2.Methodology

The methodology is described in two main sections. In the first, the computational code (i.e., BEM theory) is presented for aerodynamic analysis of the wind turbine blade, and in the other optimization and details of the integration of the optimization algorithm and BEM code are described.

2.1 Blade Element Momentum Theory

BEM combines blade element and onedimensional momentum theory. In this method, the wind turbine blades are divided into a number of independent elements along blade length. At each section, twodimensional force balances are applied by considering torque and thrust produced by the section. At the same time, the momentum balance for the annular element is applied. Finally, a set of equations are derived that can be solved for each blade section [18]. Because of the rotational flow, the angle of attack at each section of blade is determined as follows:

$$\alpha = \alpha_{\text{pitch}} - \varphi \tag{1}$$

 $\boldsymbol{\varphi}$ is inflow angle of attack that is calculated from the below:

$$\varphi = \arctan \frac{V_a}{V_{rot}}$$
(2)

where α_{pitch} is pitch angle, V_a and V_{rot} are induced velocities in rotor plane that are determined by means of axial and tangential induction factor, as follows:

$$\mathbf{V}_{\mathbf{a}} = (1 - \mathbf{a})\mathbf{V}_{\mathbf{0}} \tag{3}$$

$$V_{\rm rot} = (1 + a')\Omega r \tag{4}$$

The normal and tangential force coefficients are determined by projecting lift and drag coefficients in those directions.

$$C_{\rm N} = C_{\rm L} \cos(\varphi) + C_{\rm D} \sin(\varphi)$$
⁽³⁾

$$C_{T} = C_{L} \sin(\phi) - C_{D} \cos(\phi)$$
⁽⁶⁾

Finally, the following relations are obtained by the use of mathematical computation for axial and tangential induction factors:

$$a = \frac{1}{\frac{4\sin^2\varphi}{\sigma C_N} + 1} \tag{7}$$

$$a' = \frac{1}{\frac{4\sin\phi\cos\phi}{\sigma c_{\rm T}} - 1} \tag{8}$$

where σ is solidity and is defined as the fraction of the annular area in the control volume that is covered by the blades. Prandtl [18] derived a correction factor F to the axial and tangential induction factor to correct the assumption of an infinite number of blades. With regard to the initial BEM assumptions, for each section of blade the following algorithm is applied independently:

- 1. Initial guess for a and a'
- 2. Calculate inflow angle (Eq. 2)
- 3. Calculate local angle of attack (Eq. 1)
- 4. Read lift and drag coefficient from tabulated data

- 5. Calculate axial and tangential induction factor (Eq. 7 and 8)
- 6. If error of a and a' are more than tolerance considered, go to step 2, otherwise go to next step
- 7. Calculate aerodynamic forces

2.2 Optimization

In this section, we try to formulate the aerodynamic optimization of the blade. Optimization problems are made up of a set of design variables, an objective function and a set of constraints. The focus of optimization is to determine the values of the design variables that lead to either minimizing or maximizing the objective function while satisfying the constraints. From this point of view, the nonlinear, constrained optimum design problem can be expressed mathematically as follows:

Find
$$x = [x_1, x_2, x_3, \dots, m]^T$$

To minimize $f(x)$

Subjected to

$$g_j(x) \le 0$$
 $j = 1, 2, ... m$
 $x_i^L \le x_i \le x_i^U$ $i = 1, 2, ... n$ (9)

where x represents the design variable vector, n is the number of design variables and m is the number of constraints. f(x) is the objective function which depends on the values of the design variables. $g_j(x)$ denotes the inequality constraint. \mathbf{x}_i^L and \mathbf{x}_i^U are the lower and upper limits of the design variables, respectively, and they simply limit the region of search for the optimization.

Objective function

The output power is one of the important objectives for aerodynamic optimization of the wind turbine blade. In the current work, output power is applied as the outputs of BEM code for calculating the objective functions.

Design variables

Distribution of the chord and twist along the blade length is shown in the following equations.

$$C = ar + b \tag{10}$$

$$T = ce^{dr} + e \tag{11}$$

where r is distance from root and a, b, c, d and e are coefficients. Placement of the airfoils along the blade length of the wind turbine in this case study is illustrated in Fig. 1. Each type of airfoil is placed at a specific distance from root of blade.

In this study, chord and twist rates and the placement of the airfoils along blade length are considered as a design variables, and are shown in Table 1.

Constraint

In the optimization, lower and upper value limitations for twist and chord rate and the placement of the airfoils along the blade length are considered as constraints. Chord and twist constraints are shown in Figs 2 and 3. Additionally, a lower limitation (300) is considered for output power.

2.2.1 Optimization Method

Genetic algorithms are based on natural selection mechanisms and use only the values of the objective function. In the other words, they do not use gradients of the objective function. The genetic algorithm is based on a collection or population of candidate solutions to the optimization problem. The algorithm works in an iterative manner. At each iteration, three operators (selection, crossover and mutation) are applied to the entire population of designs. The iteration and the population values used for this paper are set to 100 and 500, respectively.



Fig. 1. Placement of the airfoils along the blade distribution

Notation	Design variables	Lower bounds	Upper bounds	Unit
<i>x</i> ₁	coefficient (a)	-0.1	-0.09	-
<i>x</i> ₂	coefficient (b)	2	3	-
<i>x</i> ₃	coefficient (c)	27	33	-
<i>x</i> ₄	coefficient (d)	-0.19	-0.13	-
<i>x</i> ₅	coefficient (e)	-1	1	-
<i>x</i> ₆	Placement of the FFA-W3	10	14	m
x ₇	Placement of the MIX	14	16	m
<i>x</i> 8	Placement of the NACA 63-600	19	21	m
<i>x</i> 9	Placement of the NACA 63-450	21	22.5	m

Table 1. Design variables of the problem



Fig. 2. Lower and upper limitation of chord distribution



Fig. 3. Lower and upper limitation of twist distribution

2.3. Integration of GA Algorithm and BEM Code

The initial value of the design variables is inputted to the BEM code from the optimization program; then, the output power is evaluated as an output of the BEM code. The GA algorithm modifies the design variables in the input file and runs the BEM code again. This process continues until optimization convergence conditions are satisfied. The flowchart of this process is shown in Fig. 4.

3.Result and discussion

3.1. Validation

To validate the aerodynamic analysis of the wind turbine, the only available data are manufacturer's measurements that are reported in [19]. In order to show the accuracy of the BEM results, output power is compared with reported data in different wind speeds. Rated speed, through which maximum output power is determined, is one of the most important wind speeds, and for this turbine it is almost equal to 14 m/s. Figure 5 shows the comparison between BEM results and reported data for different wind speeds. According to the BEM results, there is a 5 % average error for output power.

3.2. Optimum design point

Unlike in most recent studies, the challenge that has been discovered in the aerodynamic analysis and optimization process here relates to the complexity of geometry. In spite of this difficulty, the results of optimization can satisfy the constraints well.

The optimum design variables are



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Fig. 4. Integrated diagram of GA and BEM code



Fig. 5. Output power of modified BEM results and brochure data

demonstrated in Table 2. We can observe that most of the optimal design points are between their lower and upper bound. This table illustrates that the optimum values for chord distribution are -0.091969 and 2.463606, and for twist are 27.111365, -0.188605 and -0.319621. The optimal placements of airfoils (FFA-W3, MIX, NACA 63-600, NACA 63-450) are 13.2706, 20.7983 14.9901, and 22.3966 m, respectively.

The optimal chord and twist distributions are shown in Figs. 6 and 7. Figure 6 shows that the chord distribution of the optimal design is almost identical to the preliminary design. According to these figures, the twist rate variation in the optimized shape is more than that of the chord rate along the blade. Figure 7 shows that the difference between the new design twist and the old one is higher near the root than in the rest of the blade length. Figure 8 illustrates the placement of the airfoils along the blade length.

Finally, the output power is improved by 10 % in the optimum design point compared to the base design at rated speed.

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Table 2. Optimum values of design variables and objective function						
Design variables	Lower bounds	Upper bounds	Optimal			
<i>x</i> ₁	-0.1	-0.09	-0.091969			
<i>x</i> ₂	2	3	2.463606			
<i>x</i> ₃	27	33	27.111365			
<i>x</i> ₄	-0.19	-0.13	-0.188605			
<i>x</i> ₅	-1	1	-0.319621			
<i>x</i> ₆	10	14	13.2706			
x ₇	14	16	14.9901			
<i>x</i> ₈	19	21	20.7983			
<i>x</i> 9	21	22.5	22.3966			



Fig. 6. Optimal chord rate



Fig. 7. Optimal twist rate



Fig. 8. Optimal placement of the airfoils along the blade length

4. Conclusion

The aerodynamic optimization of the wind turbine blade is investigated by integrated GA algorithm and modified BEM code. In order to improve output power in the same blade length, the GA algorithm was chosen as an optimization method, and for this purpose chord and twist distribution and placement of airfoils are considered as design variables. Aerodynamic analysis of the wind turbine was also carried out by the use of BEM theory, which is a well-known method in aerodynamics of rotary wings, especially in wind turbines. One of the challenges facing this study related to the aerodynamic analysis of the case study. Unlike most aerodynamic studies, which are of wind turbines with a simple geometry, this turbine has been designed with more complexity, so that it included different types of airfoils that were also twisted and tapered along the blade length. Never the less, the BEM code determined acceptable results compared to the manufacturer's measurements. Results show that by choosing optimized variables, output power can be increased by 10 % compares to the base design in a specific wind speed.

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